

# Hybrid Cascaded MPC UAV System and Corresponding Waypoint Tracking Simulation

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## 1 Mathematical Formulation of Cascaded Hybrid MPC System

We formulate cascaded hybrid UAV system as: a high-level Model Predictive Controller (MPC) acts as a kinematic trajectory generator to produce a continuously moving virtual target, which is subsequently tracked by a low-level PID controller.

### 1.1 Hybrid State Space

The overall system encompasses continuous physical states, the virtual target state from the MPC, and discrete logic states.

- **Continuous Physical States:** Position  $p \in \mathbb{R}^3$ , velocity  $v \in \mathbb{R}^3$ , orientation  $\Theta \in \mathbb{R}^3$ , and angular velocity  $\omega \in \mathbb{R}^3$ .
- **Continuous Virtual State:** Virtual target position  $p_v \in \mathbb{R}^3$  driven by the MPC.
- **Discrete State:** The index of the current waypoint target  $q \in \mathcal{Q} = \{0, 1, 2, \dots, N\}$ .

The complete hybrid state vector  $\chi$  is defined as:

$$\chi = \begin{bmatrix} p \\ v \\ \Theta \\ \omega \\ p_v \\ q \end{bmatrix} \in \mathbb{R}^{15} \times \mathcal{Q} \quad (1)$$

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## 1.2 High-Level Kinematic Model Predictive Control (Discrete)

To prevent aggressive attitude oscillations, the MPC solves a finite-horizon Optimal Control Problem at discrete control intervals  $k$  to smoothly advance the virtual target  $p_v$  towards the absolute waypoint  $W_q$ .

The discrete-time linear prediction model for the virtual target is formulated as:

$$p_v(k+1) = Ap_v(k) + Bu_{mpc}(k) \quad (2)$$

where  $A = I_{3 \times 3}$ ,  $B = I_{3 \times 3} \Delta t$ , and  $u_{mpc} \in \mathbb{R}^3$  is the virtual velocity command.

At each step  $k$ , the MPC minimizes the following quadratic cost function over a prediction horizon  $H_p$ :

$$\min_{U_{mpc}} \sum_{i=0}^{H_p-1} (\|p_v(k+i) - W_q\|_Q^2 + \|u_{mpc}(k+i)\|_R^2) \quad (3)$$

Subject to box constraints on the input velocity:

$$u_{min} \leq u_{mpc}(k+i) \leq u_{max}, \quad \forall i \in \{0, \dots, H_p - 1\} \quad (4)$$

where  $Q$  and  $R$  are positive definite weight matrices penalizing tracking error and control effort, respectively. The first element of the optimal sequence  $u_{mpc}^*(k)$  is applied to update the virtual target.

## 1.3 Low-Level Continuous Dynamics (Flow Map $f$ )

Between discrete waypoint transitions, the system undergoes continuous-time physical evolution governed by the flow set  $C$  and flow map  $f(\chi)$ .

The flow set dictates that continuous flight is maintained as long as the UAV has not reached the tolerance radius  $\epsilon$  of the current waypoint, or it is at the final waypoint:

$$C = \left\{ \chi \in \mathbb{R}^{15} \times \mathcal{Q} \mid \|p - W_q\| \geq \epsilon \vee q = N \right\} \quad (5)$$

Within  $C$ , the continuous dynamics are decoupled into translational and rotational components to match the state dimensions:

$$\dot{\chi} = f(\chi) = \begin{bmatrix} \dot{p} \\ \dot{v} \\ \dot{\Theta} \\ \dot{\omega} \\ \dot{p}_v \\ \dot{q} \end{bmatrix} = \begin{bmatrix} v \\ \mathcal{F}_{trans}(p, v, \Theta, \omega, u_{pid}([p, v, \Theta, \omega], p_v)) \\ W(\Theta)\omega \\ \mathcal{F}_{rot}(\Theta, \omega, u_{pid}) \\ u_{mpc}^*(t) \\ 0 \end{bmatrix}, \quad \chi \in C \quad (6)$$

where  $\mathcal{F}_{trans}$  represents the translational dynamics (acceleration bounded by thrust and gravity), and  $\mathcal{F}_{rot}$  governs the rotational kinematics and attitude dynamics, both controlled by the PID loop.

It is worth noting that the evolution of the virtual target, denoted as  $\dot{p}_v = u_{mpc}^*(t)$ , is formulated in continuous-time (Newton's notation) to close the theoretical hybrid system flow map. In the actual digital simulation environment, this continuous derivative is realized through first-order Euler numerical integration at each control timestep  $\Delta t$ :

$$p_v(k+1) = p_v(k) + u_{mpc}^*(k)\Delta t \quad (7)$$

This discrete algebraic update seamlessly bridges the low-frequency optimal velocity commands generated by the discrete MPC solver with the high-frequency continuous physical tracking executed by the underlying PID controller.

#### 1.4 Low-Level Discrete Event Transitions (Jump Map $g$ )

The discrete transitions (waypoint switching) are triggered strictly when the physical position  $p$  enters the  $\epsilon$ -neighborhood of the current target  $W_q$ . The jump set  $D$  is defined as:

$$D = \left\{ \chi \in \mathbb{R}^{15} \times \mathcal{Q} \mid \|p - W_q\| < \epsilon \wedge q < N \right\} \quad (8)$$

Upon intersecting  $D$ , the system state undergoes an instantaneous jump defined by the map  $g(\chi)$ . Physical states remain continuous, while the discrete target index advances:

$$\chi^+ = g(\chi) = \begin{bmatrix} p \\ v \\ \Theta \\ \omega \\ p_v \\ q+1 \end{bmatrix}, \quad \chi \in D \quad (9)$$

#### 1.5 Overall Hybrid Automaton

Combining the flow and jump components, the overall cascaded system is written as the hybrid automaton  $\mathcal{H}$ :

$$\mathcal{H} : \begin{cases} \dot{\chi} = f(\chi), & \chi \in C \\ \chi^+ = g(\chi), & \chi \in D \end{cases} \quad (10)$$

## 2 Simulation: Waypoint Tracking

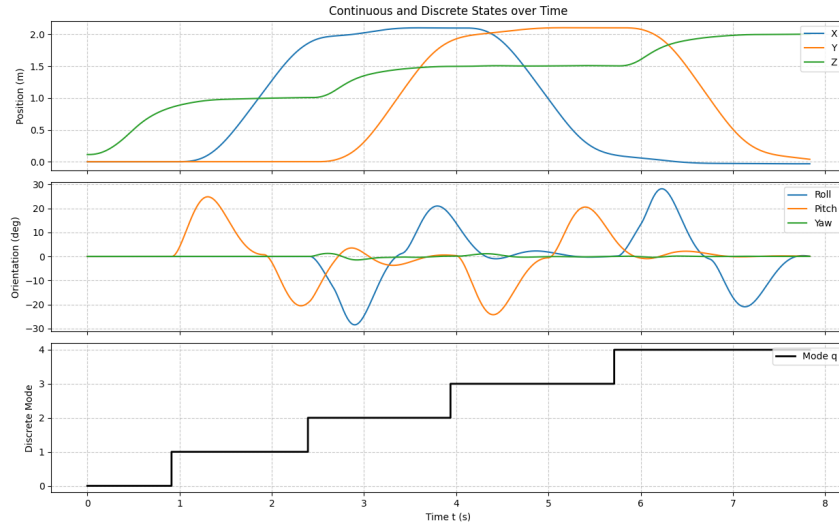


Figure 1: Waypoint Tracking Simulation: State Plot

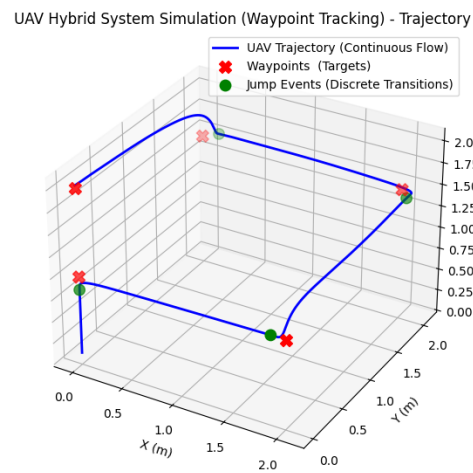


Figure 2: Waypoint Tracking Simulation: Trajectory Plot